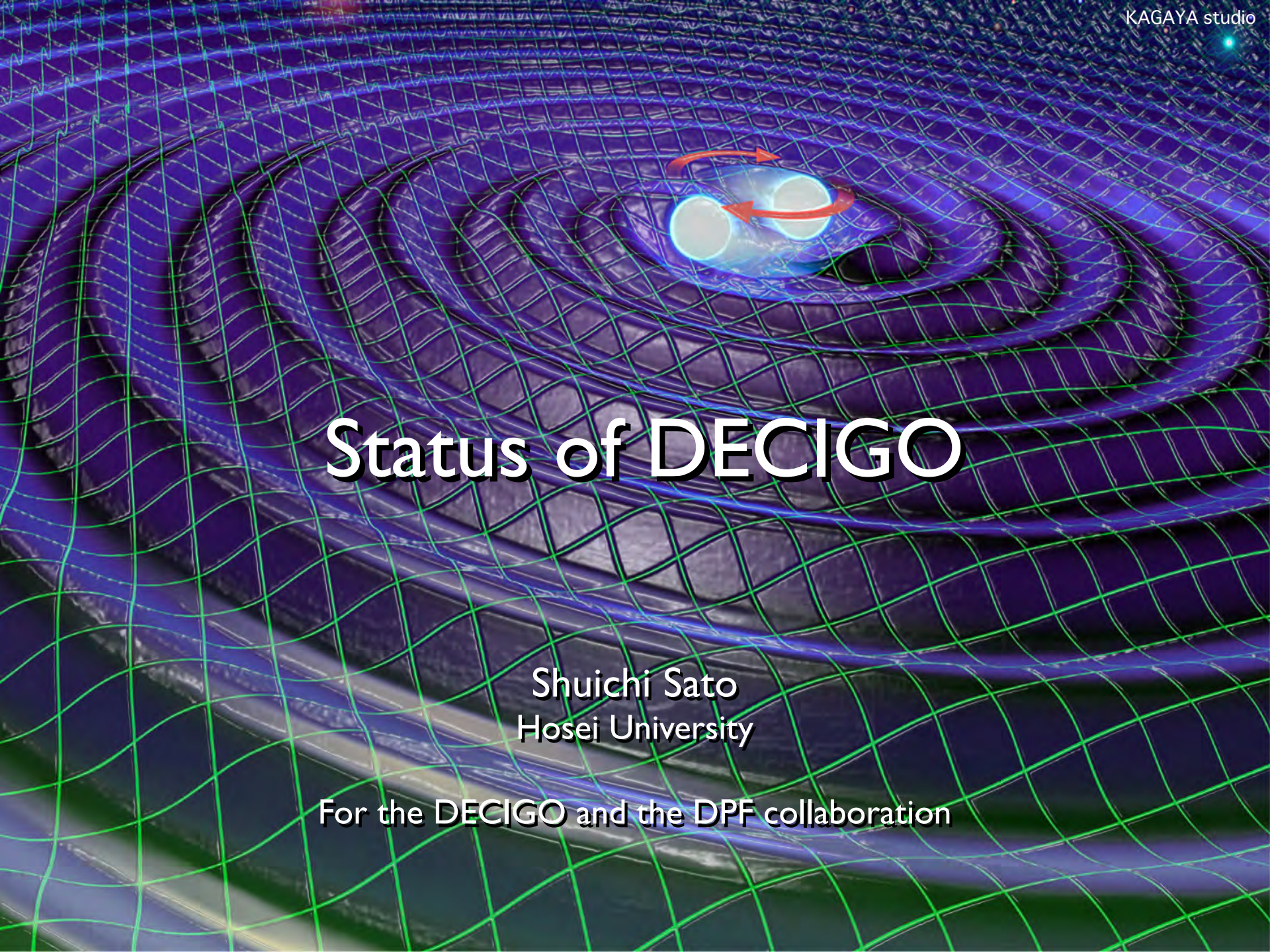


The background of the slide is a 3D visualization of a gravitational well, represented by a green grid that curves and dips into a deep, swirling shape. In the center of this well, two glowing blue spheres are shown in the process of merging. Red curved arrows around the spheres indicate their orbital paths and the direction of their inward motion.

Status of DECIGO

Shuichi Sato
Hosei University

For the DECIGO and the DPF collaboration

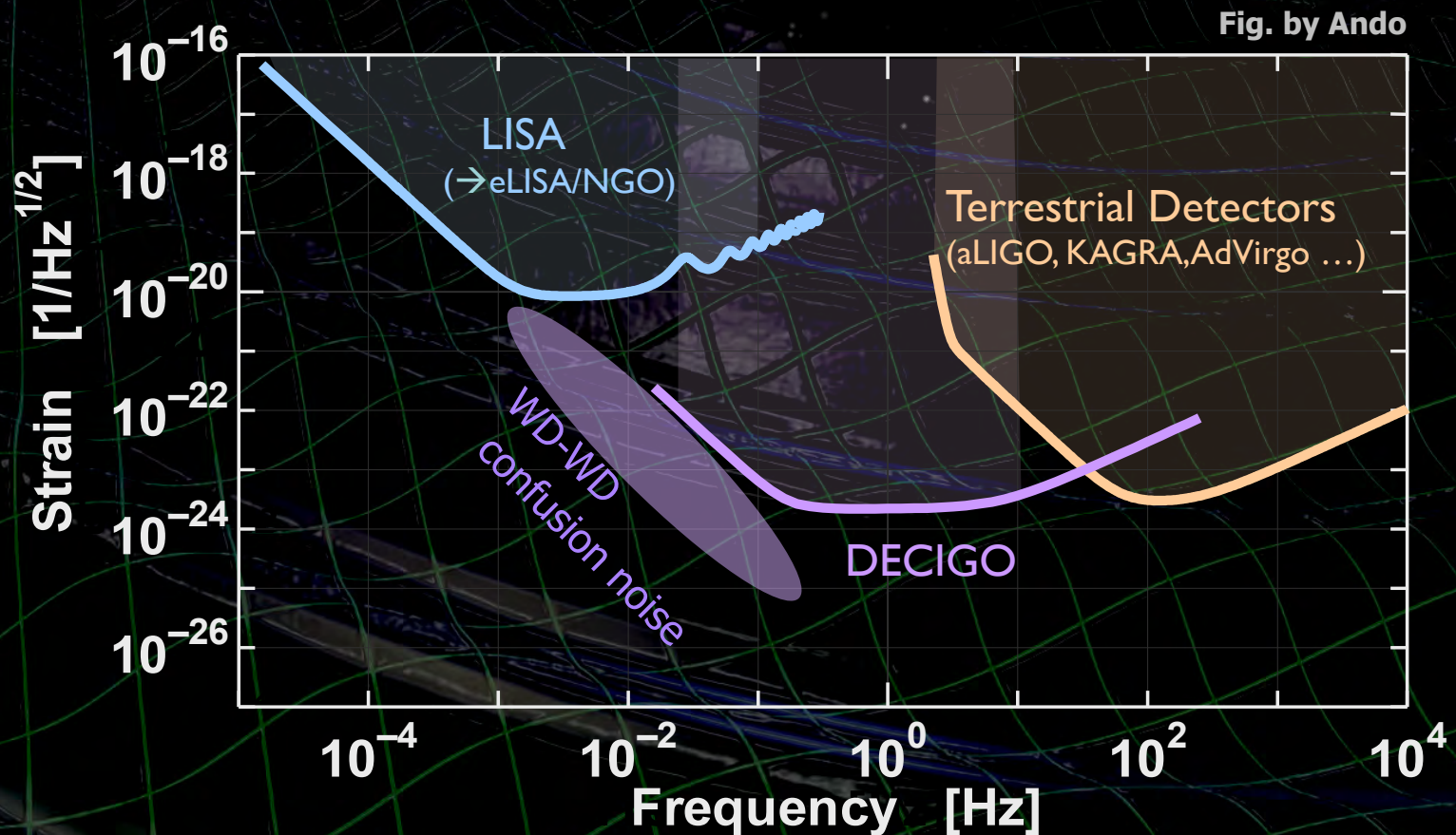
Outline

- ❖ DECIGO
- ❖ DECIGO pathfinder
- ❖ Mission status

Idea of DECIGO

❖ DECI-hertz Interferometer Gravitational wave Observatory

- ❖ Seto, Kawamura and Nakamura, *PRL*87, 221103(2001)
- ❖ Bridges the gap between LISA and terrestrial detector
- ❖ Low confusion noise → Potentially high sensitive instruments



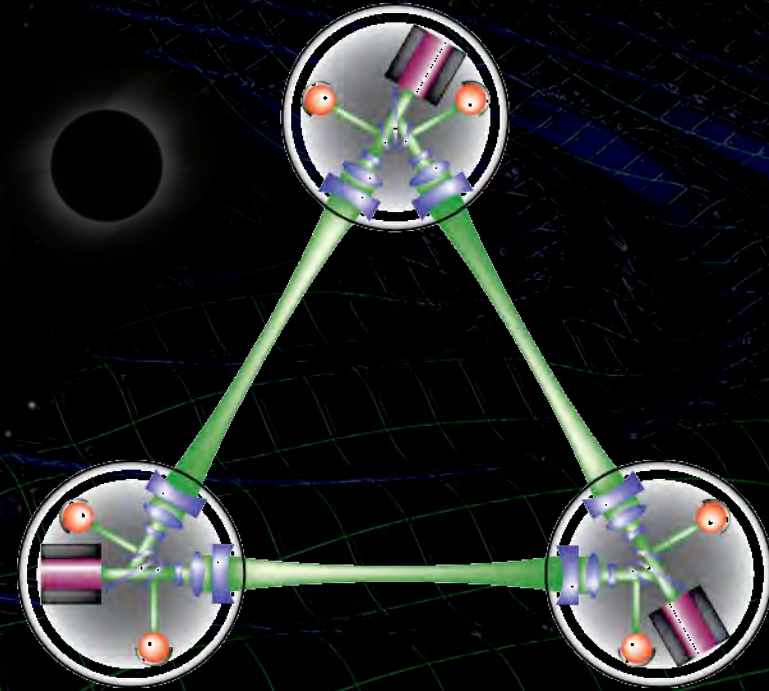
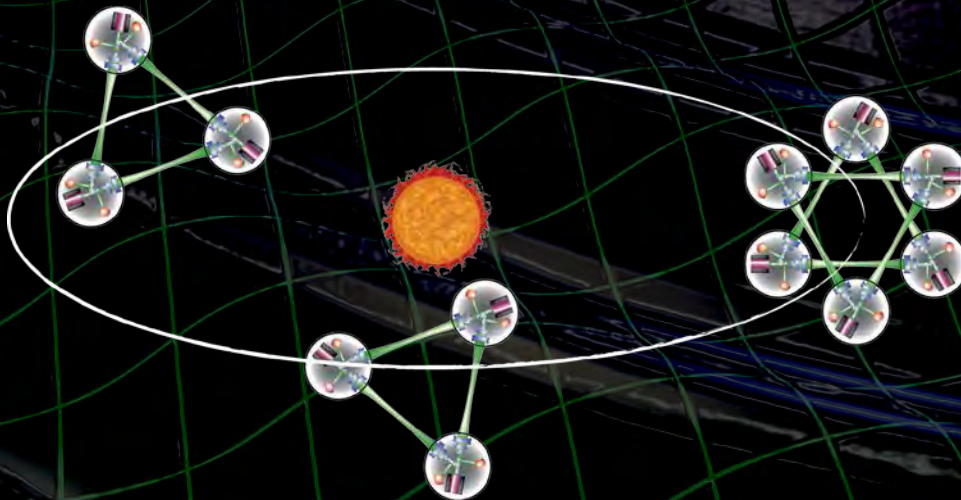
Pre-conceptual design

❖ Interferometer topology

- ❖ Differential FP interferometer
- ❖ Three interferometers for redundancy
- ❖ Drag-free controlled S/Cs

❖ Constellation

- ❖ 4 interferometer units
- ❖ 2 overlapped units → Cross correlation
- ❖ 2 separated units → Angular resolution



Arm length:	1000 km
Mirror diameter:	1 m
Mirror mass:	100 kg
Laser wavelength:	532 nm
Laser power:	10 W
Finesse:	10

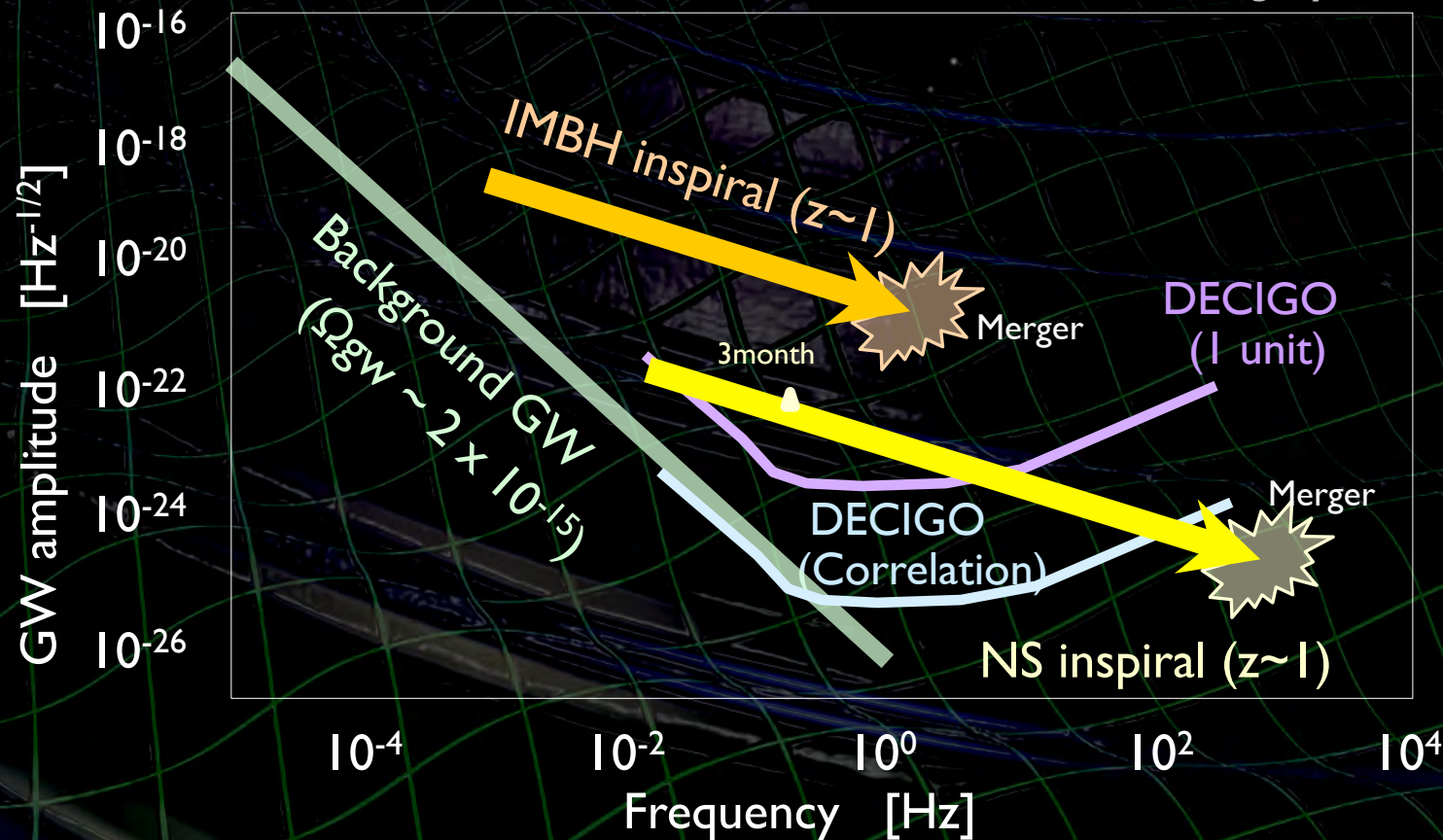
Science of DECIGO

- ❖ IMBH binary inspiral
- ❖ NS binary inspiral
- ❖ Stochastic background



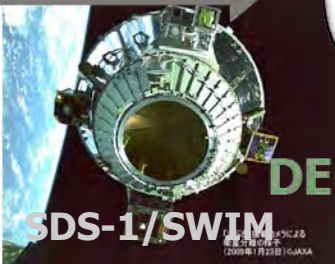
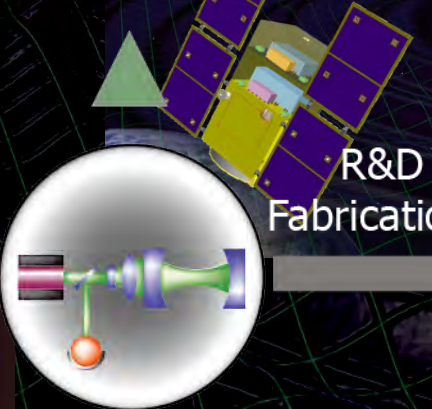
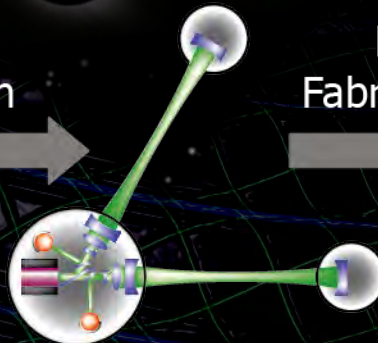
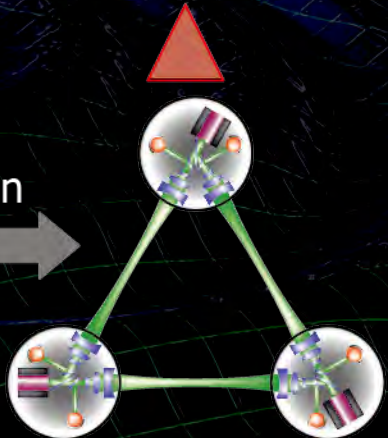
- ❖ Galaxy formation (Massive BH)
- ❖ Cosmology (Inflation, Dark energy)
- ❖ Fundamental physics

Fig. by Ando



Roadmap

Fig. by Kawamura, rev.

	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	
Mission	 SDS-1/SWIM  DECIGO Pathfinder (DPF)  Pre-DECIGO  DECIGO R&D Fabrication → R&D Fabrication → R&D Fabrication																				
Objectives	Test of key tech. in orbit GW detection Earth gravity observation									Detection of GW w/ min. spec. FP cavities between S/Cs						GW astronomy					
Scope	Single small S/C Short FP interferometer									3 S/Cs, single IFO Single unit						3 S/Cs, 3 IFOs 3 or 4 units					

Key technologies for DECIGO

❖ Precision measurement in orbit

- ❖ laser interferometry : Fabry-Perot cavity
- ❖ All dof drag-free control
- ❖ Demonstration by DPF

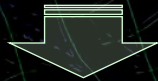
❖ Precision formation flying technique

- ❖ Long-baseline formation flight : km scale
- ❖ mm scale precision
- ❖ Demonstration by Pre-DECIGO

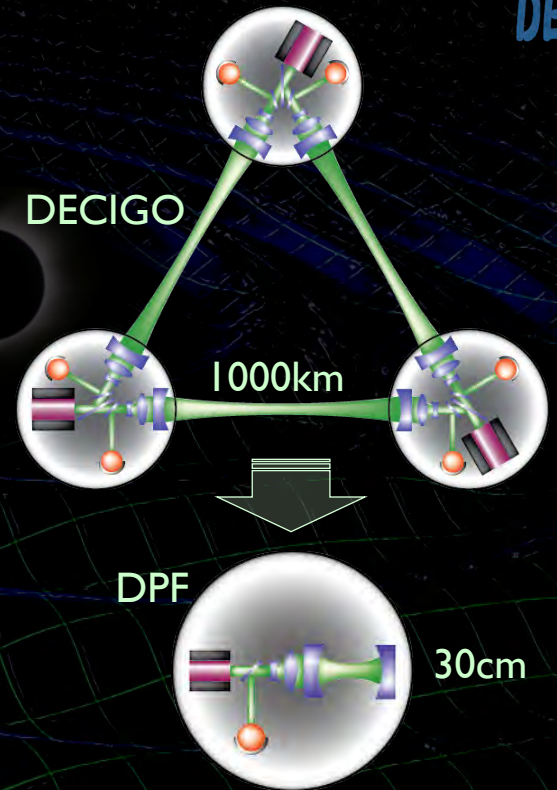
DECIGO pathfinder (DPF)

❖ First precursor satellite for DECIGO

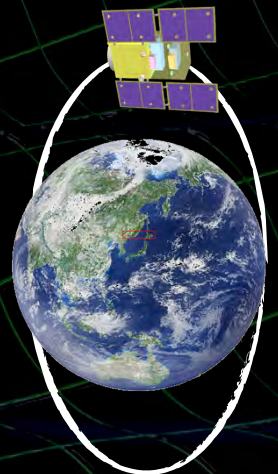
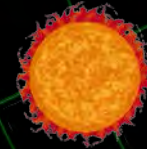
DECIGO: 1000km baseline
Formation flying S/Cs



DPF: 30cm baseline
Single satellite



- ❖ Mission scale : ~450kg
- ❖ Orbit : Low earth orbit, 500km
(sun synchronous polar orbit)
- ❖ Launch vehicle: Next-generation
Solid rocket booster (M-V FO)
- ❖ Apply for JAXA's
“Small science satellite series” program



Sensitivity of DECIGO pathfinder

Laser source : 1030nm, 25mW

IFO length : 30cm

Finesse : 100, Mirror mass : 1kg

Q-factor : 10^5 , Substrate:TBD

Temperature : 293K

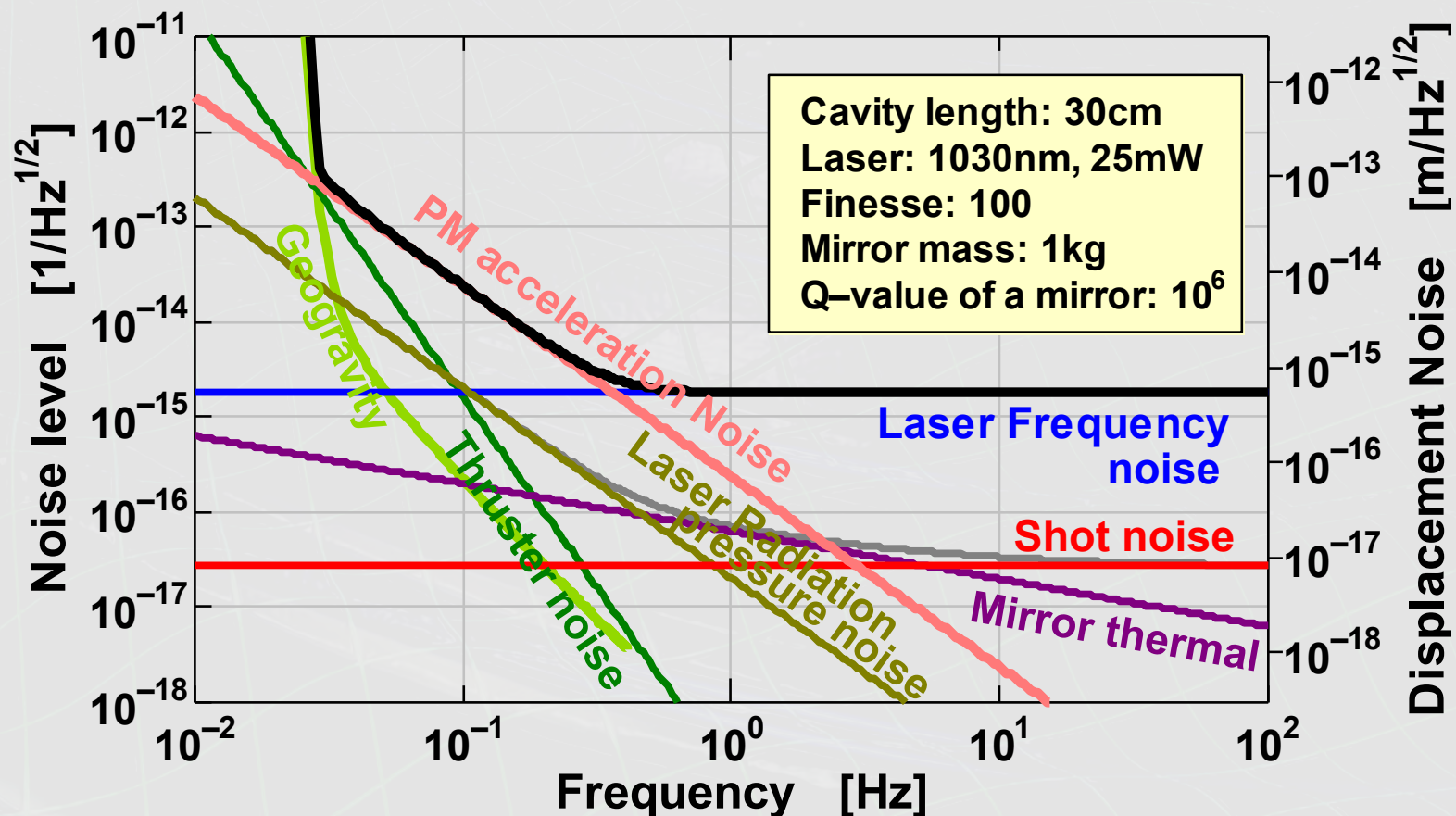
Satellite mass : 450kg

Altitude: 500km

Thruster noise: $0.1 \mu\text{N}/\text{Hz}^{1/2}$

(Preliminary parameters)

Fig. by Ando



DPF satellite overview

DPF Payload

Size : 950mm cube
 Weight : 200kg
 Power : 130W
 Data Rate: 800kbps
 Mission thruster x10

Power Supply
 SpW Comm.

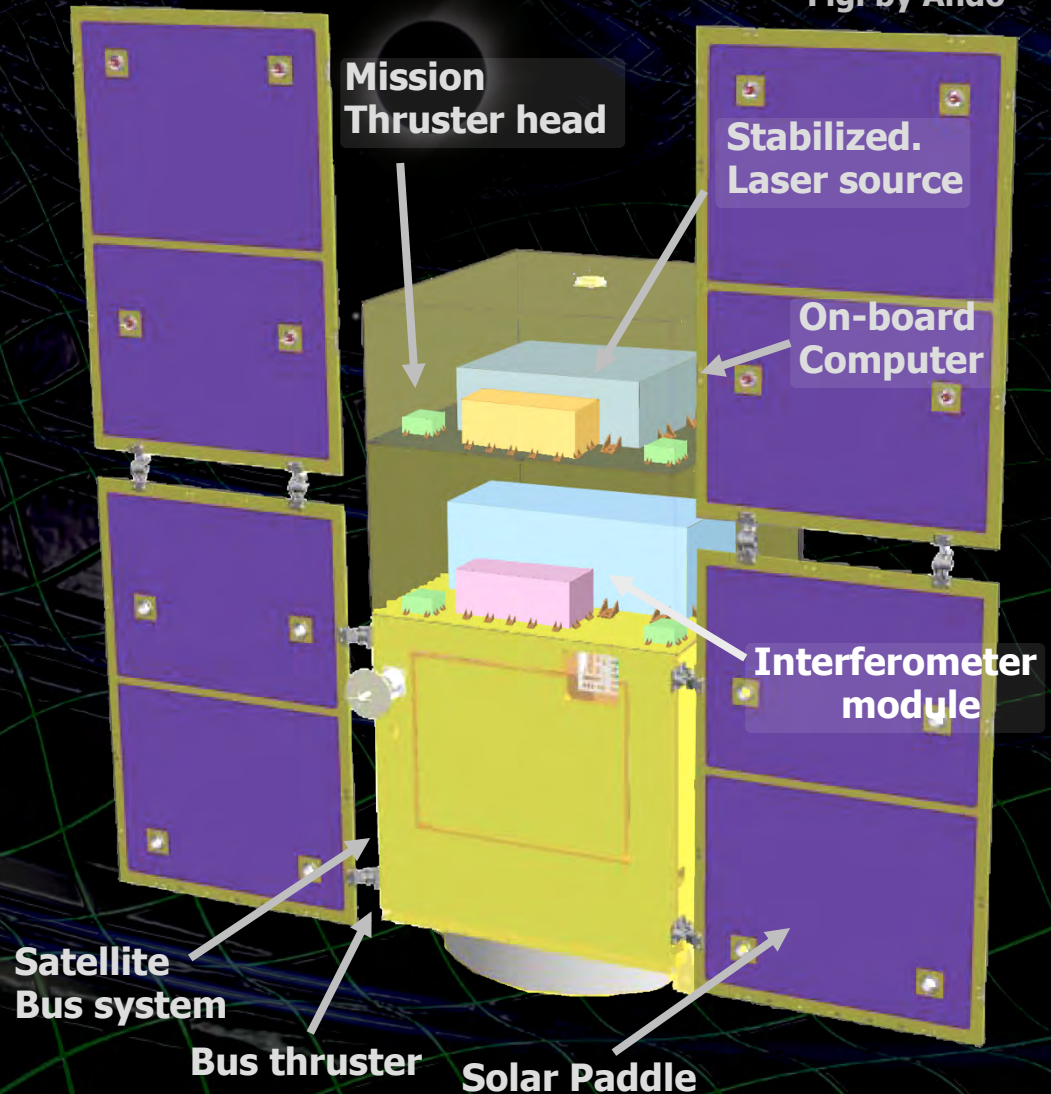


Satellite Bus

('Standard bus' system)

Size : 950x950x1100mm
 Weight : 250kg
 SAP : 960W
 Battery: 50AH
 Downlink : 2Mbps
 DR: 1GByte
 3N Thrusters x 4

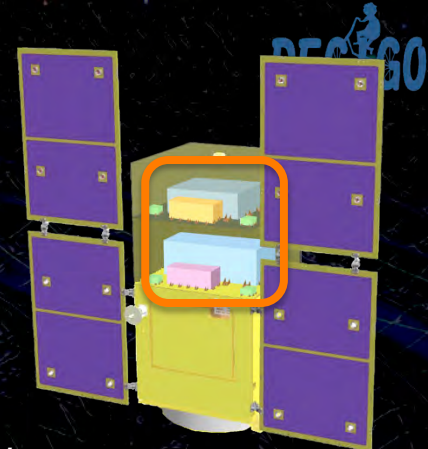
Fig. by Ando



DPF mission system (schematic)

Mission weight : ~150kg (200kg for bus)

Mission space : ~95 x 95 x 90 cm



Drag-free control

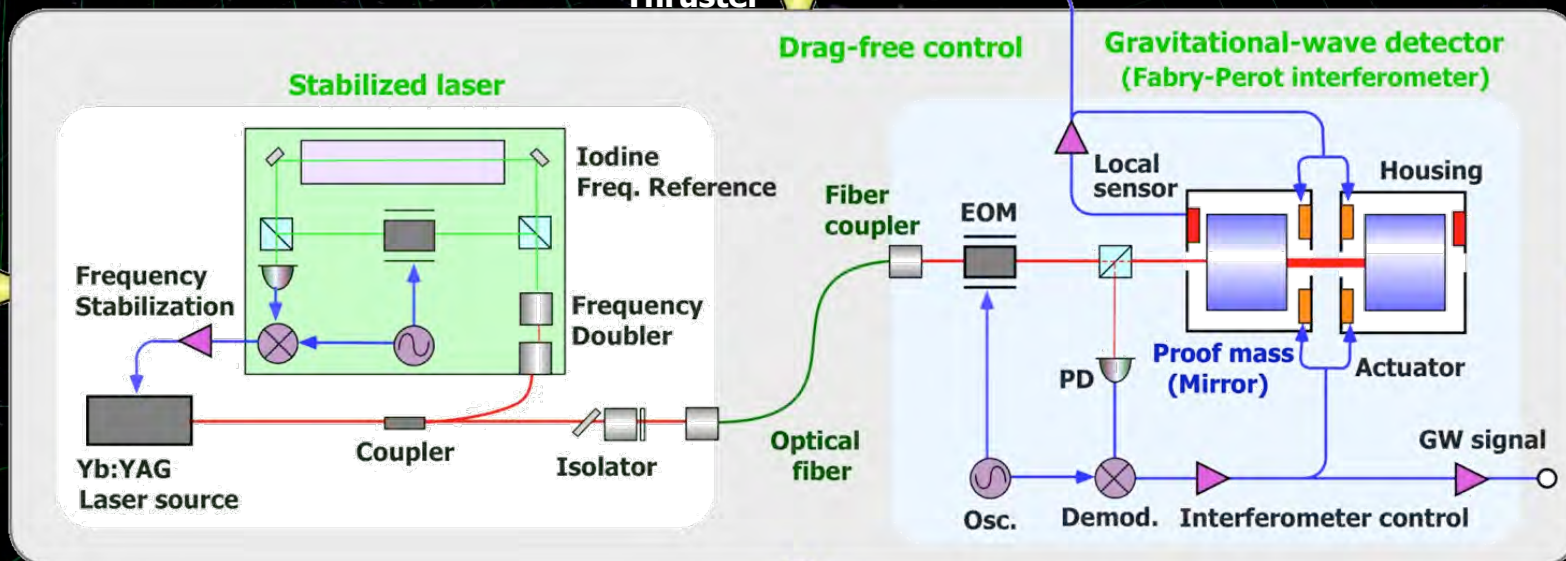
Local sensor signal

→ Feedback to thrusters

Thruster

Drag-free control

Gravitational-wave detector
(Fabry-Perot interferometer)



Stabilized laser

Yb: fiberDFB laser (1030nm)

Power : 25mW

Freq. stab. by Iodine absorption line

Fabry-Perot interferometer

Finesse : 100

Length : 30cm

Test mass : ~1kg

Signal extraction by PDH

Fig. by Ando

Mission module/system on-board

❖ Interferometer module

❖ Test mass module

*Test mass, caging frame, laser sensor,
Electro static sensor/actuator,
Test mass lock mechanism,
Charge management system*

❖ Input/output optics

*Monolithic optical bench, PDH control,
WFSs, MMT*

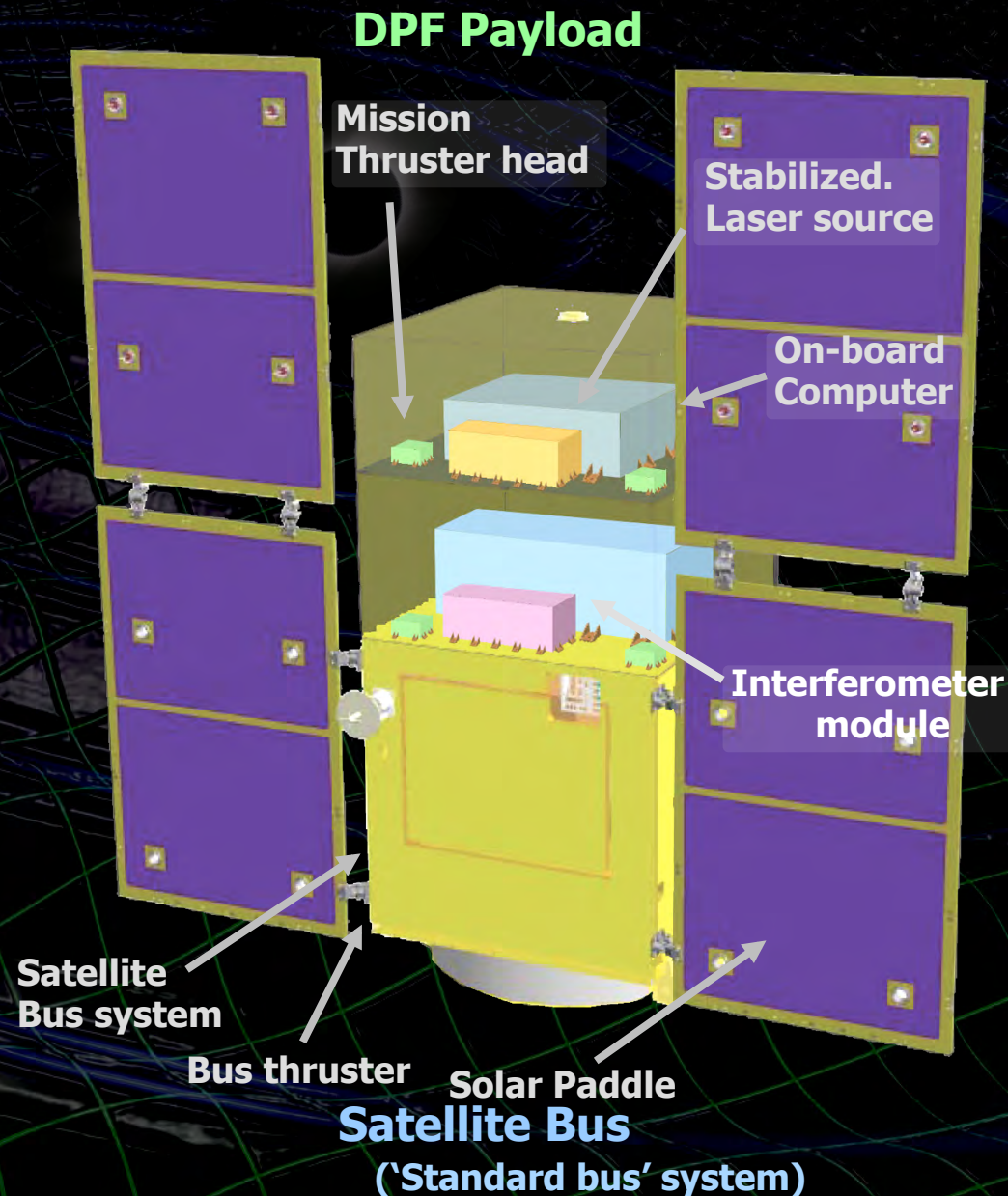
❖ Structure/shielding system

*Support frame, thermal shielding,
Vacuum enclosure*

❖ Stabilized laser module

❖ Drag-free control system

❖ Signal processing/control system



Stabilized laser module

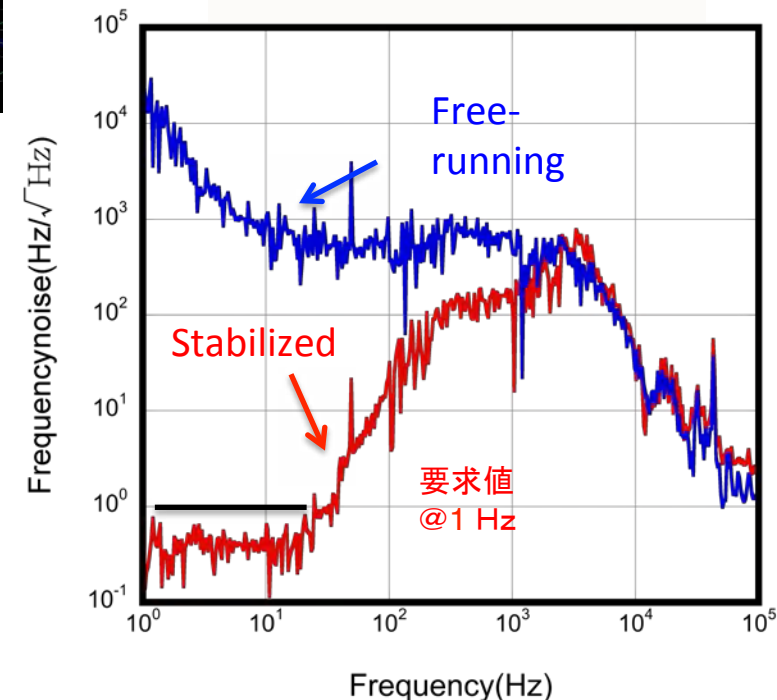
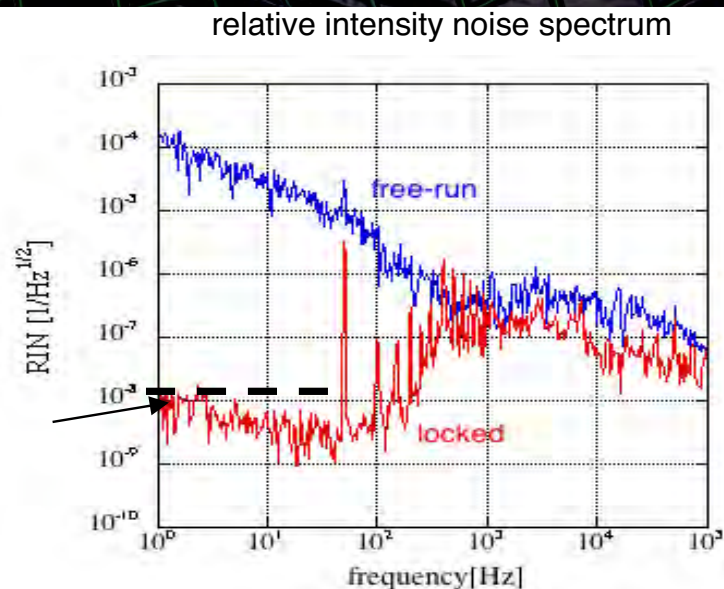
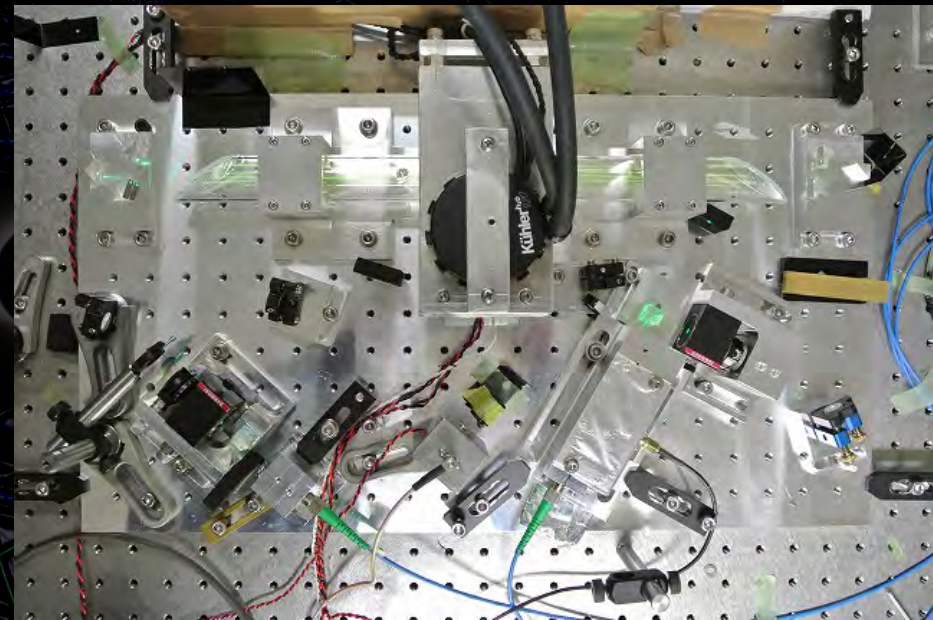
❖ Laser source

- ❖ Yb: Fiber DFB laser 1030nm (20mW)
 - ❖ YDFA amplifier (250mW)
- Compact and light, Vibration resistant*

❖ Frequency stabilization

- ❖ Iodine cell
- Vibration resistant, Long-term stability*
- ❖ Absorption line: 515nm
- ❖ Stability: 0.4 Hz/rtHz @ <10Hz

❖ BBM Size: W550 × D300 × H82 mm



Interferometer module

❖ 2 Test mass modules

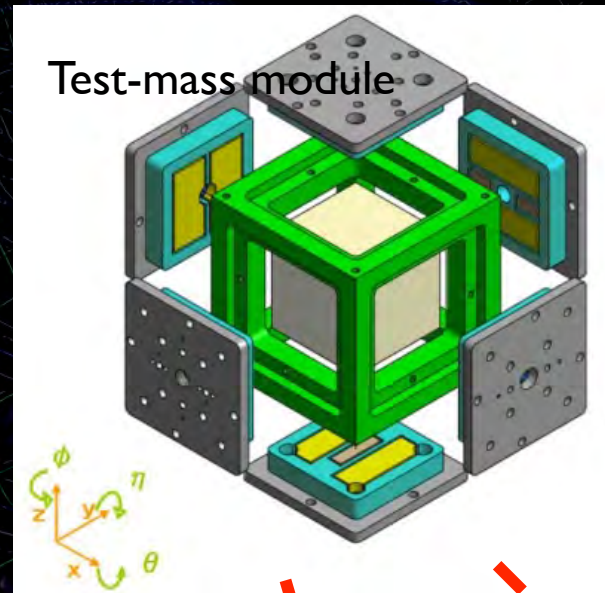
Test mass, caging frame,
Electro static sensor/actuator,
Test mass lock mechanism,
Charge management system

❖ Input/output optics

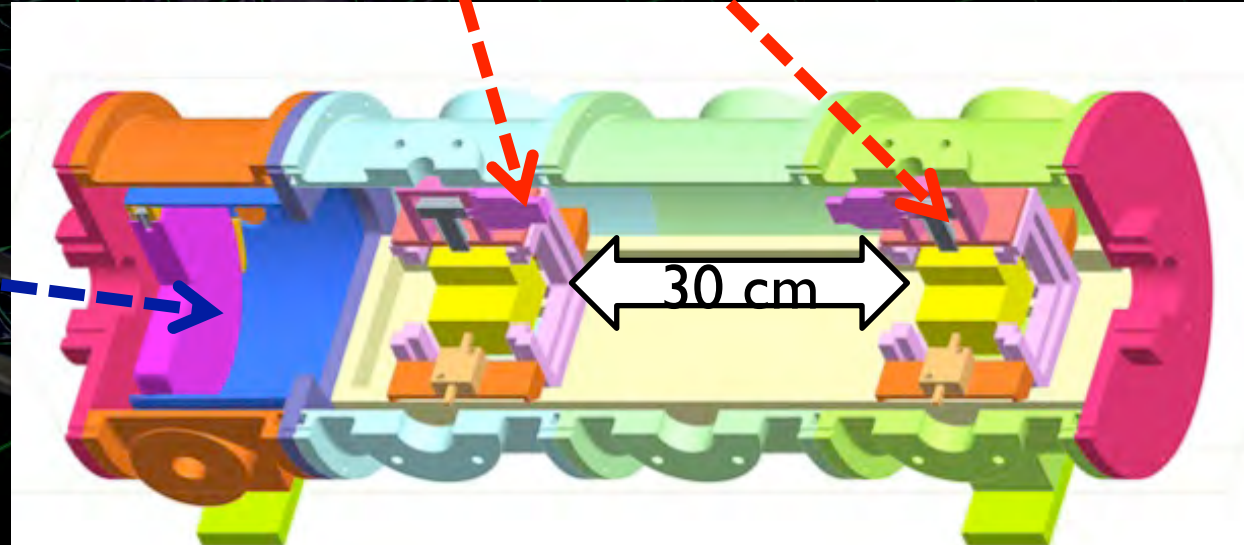
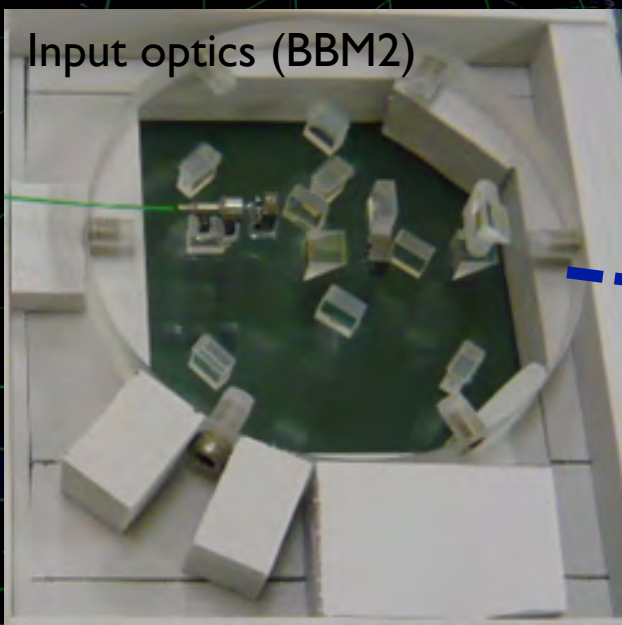
Monolithic optical bench, PDH control,
WFSs, MMT

❖ Structure/shielding system

Support frame, thermal shielding,
Vacuum enclosure (10^{-6} Pa)



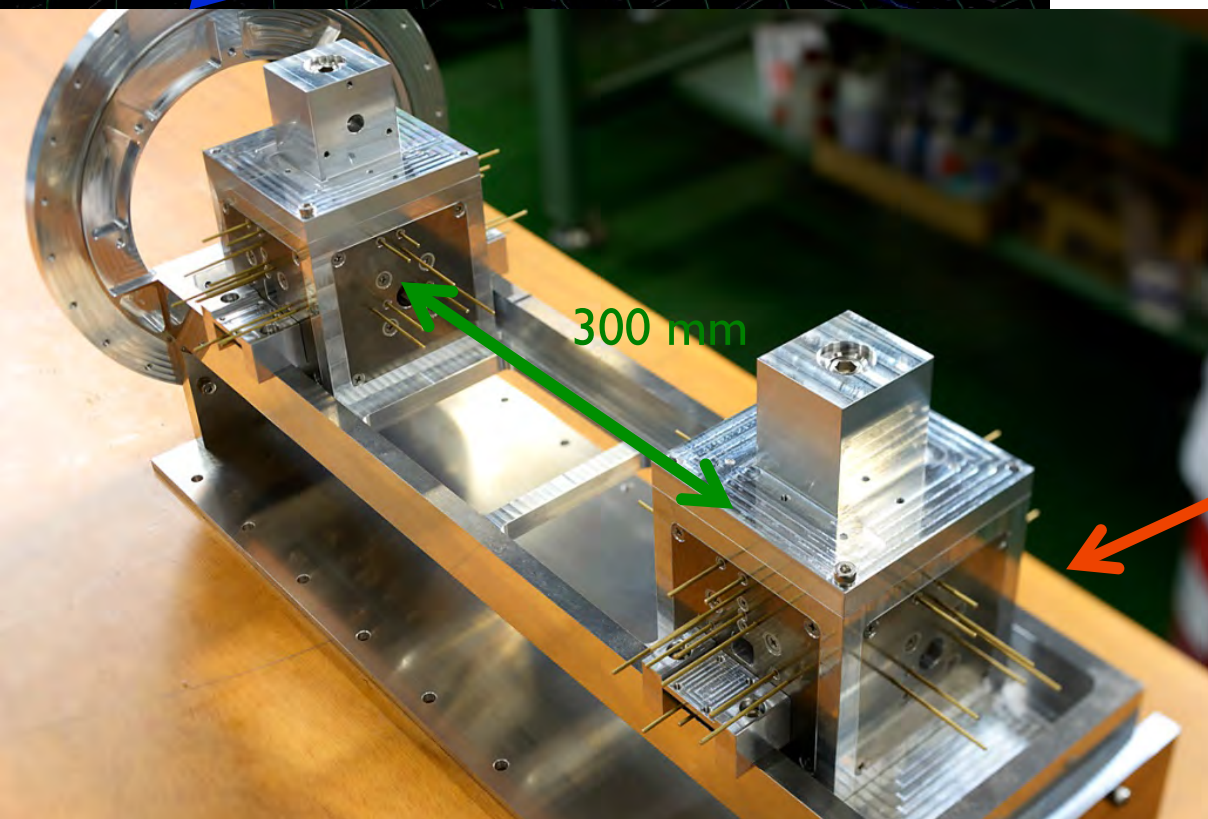
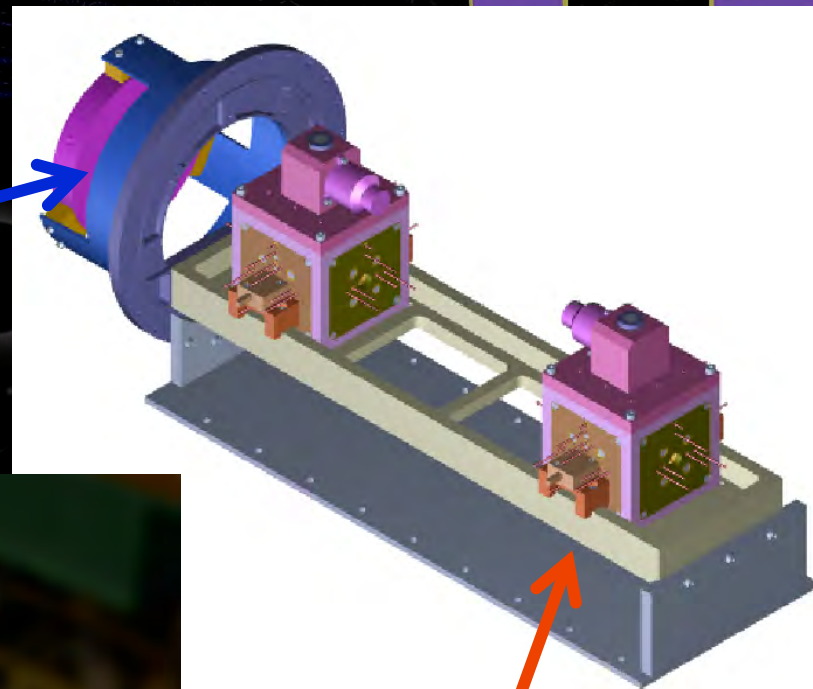
@NAOJ



Wo

Interferometer module

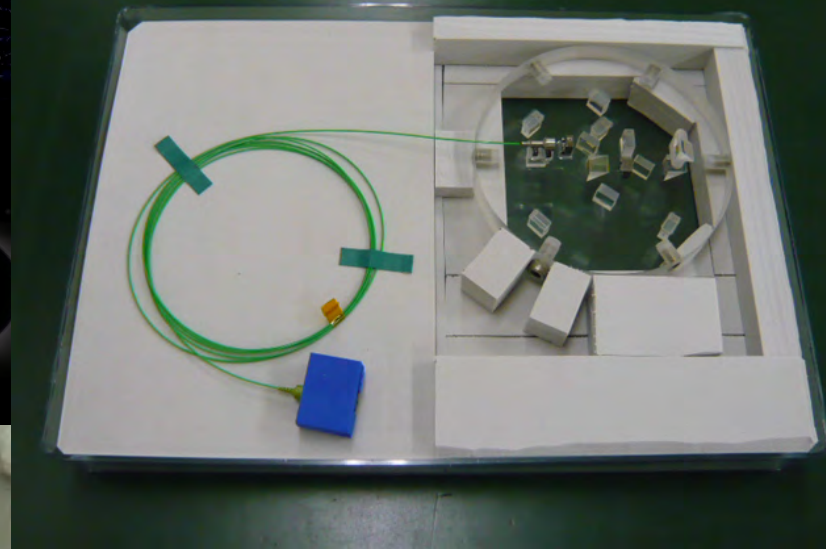
Monolithic input optics



Test mass module

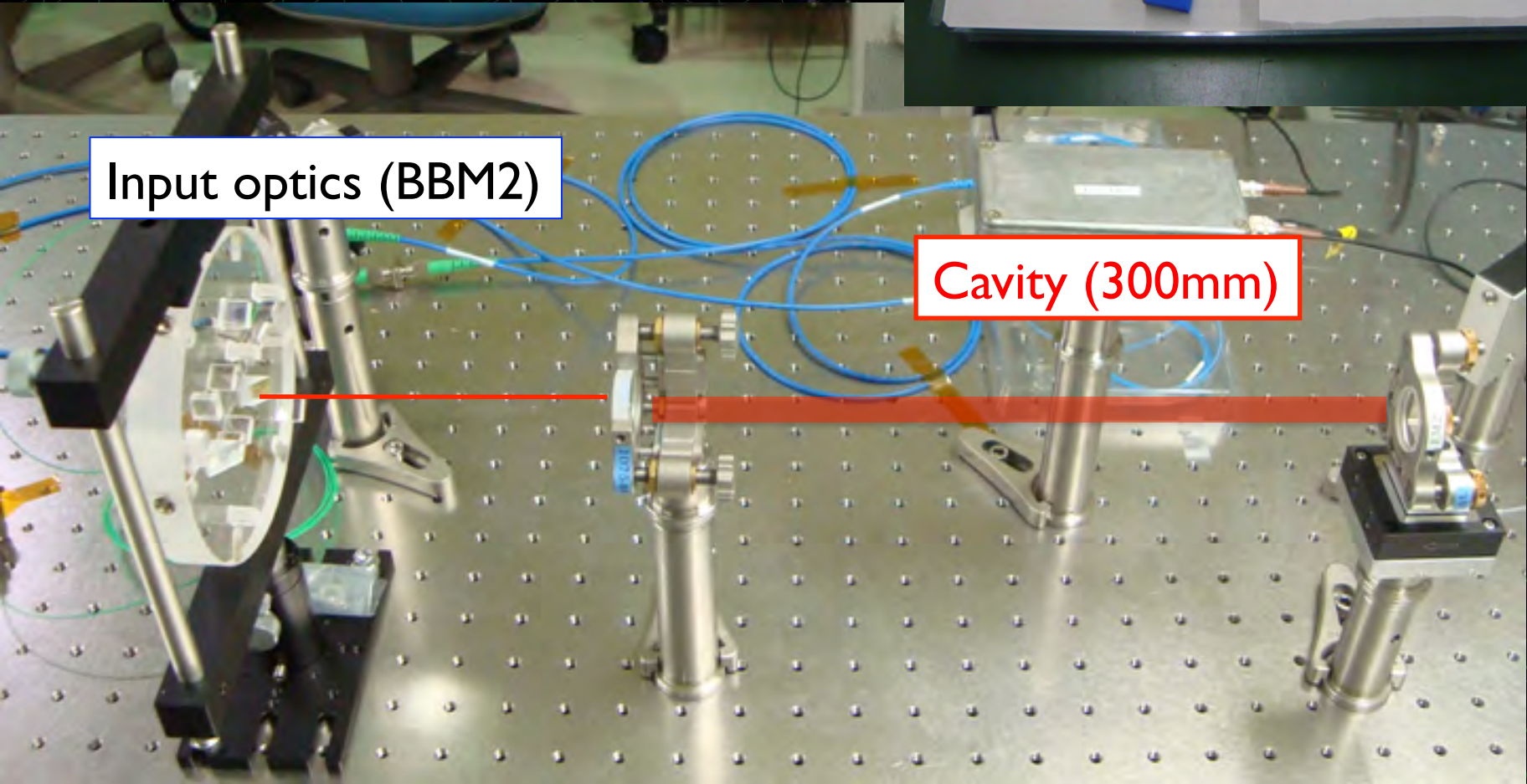
Input/output optical system

- ❖ Monolithic optical bench
- ❖ Silicate bonding
- ❖ Compacted for IFO BBM, $\varnothing 120\text{mm}$
- ❖ Optics bonded on both sides



Input optics (BBM2)

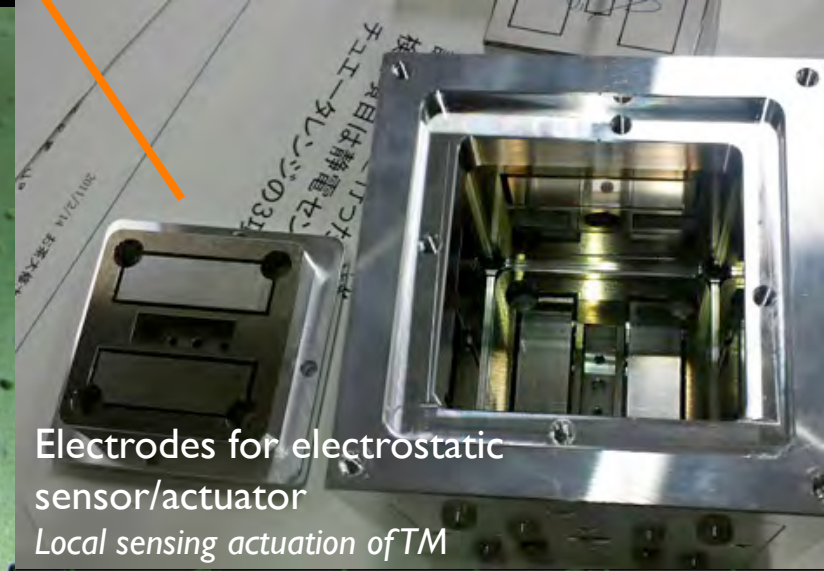
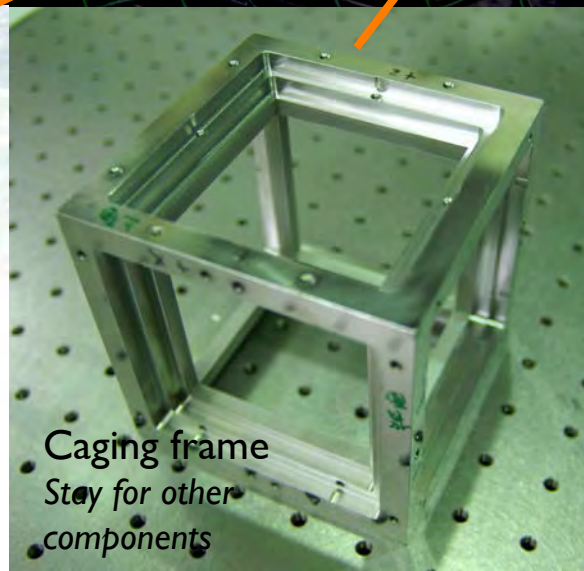
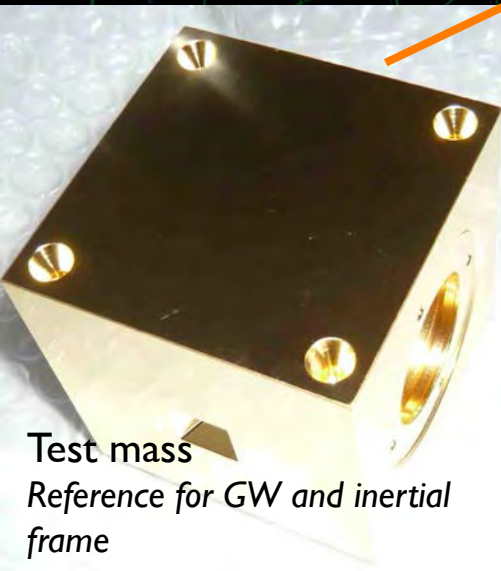
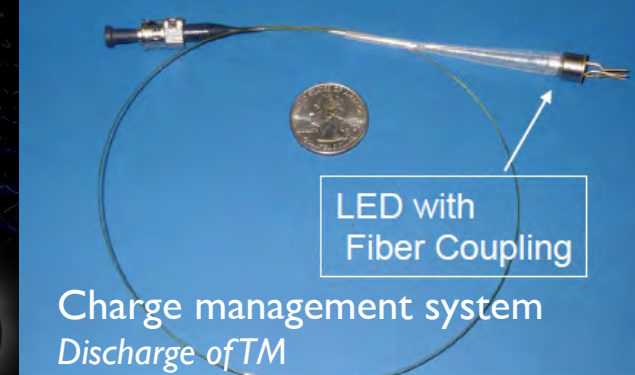
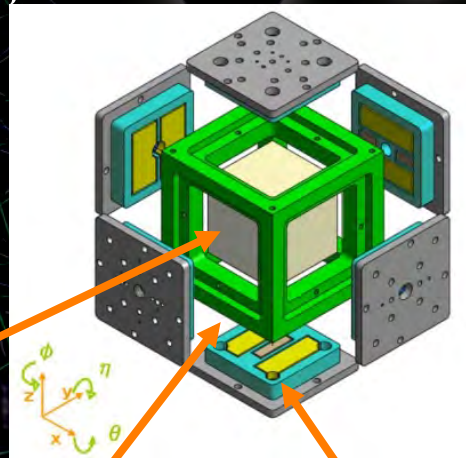
Cavity (300mm)



Test-mass module

- ❖ Test mass
- ❖ Caging frame
- ❖ Electro static sensor/actuator,
- ❖ Test mass lock mechanism,
- ❖ Charge management system

Test mass module



Drag free control

Fig. by Moriwaki

❖ Concept

- ❖ Control S/C with reference to TMs
- ❖ Diff. modes to TMs
- ❖ Comm. modes to S/C
- ❖ Using hierarchical control

❖ Disturbance source @LEO500km

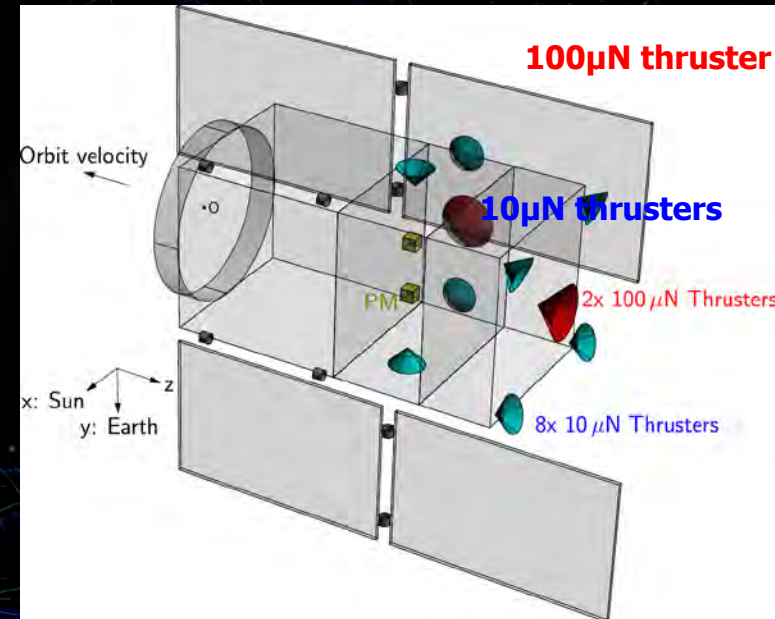
- ❖ Solar radiation pressure: $20\mu\text{N}/\text{DC}$
- ❖ Air drag: typ. $100\mu\text{N}/\text{DC}$
Annual/season dependent

❖ Micro Thruster

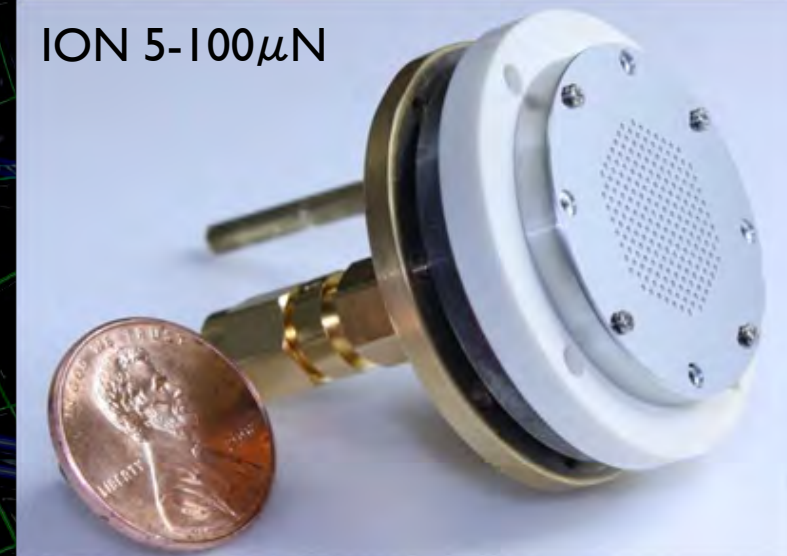
- ❖ $10\mu\text{N}$ thruster x8 for drag free control
- ❖ $100\mu\text{N}$ thruster x2 for DC drag cancellation
- ❖ FEEP, Cold gas jet, Ion thruster

❖ Attitude control

- ❖ Should work w/o thrusters
- ❖ Passive: owing to S/C shape
- ❖ Active: MTQ control



ION 5-100 μN

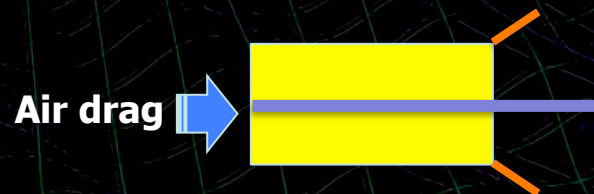


Attitude/Drag free control

❖ Passive control

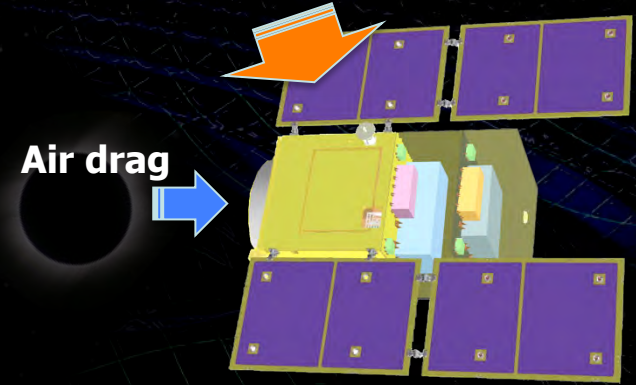
❖ “weathercock” stability (for air drag)

SAP as “wing”
Fin as “tail”



Solar radiation drag

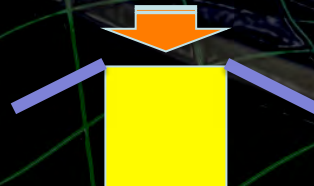
Air drag



❖ Cant for SAPs (for solar radiation drag)

Cant angle
SAP position to the S/C frame

Solar radiation drag



Signal process/control system

❖ Space Wire based system

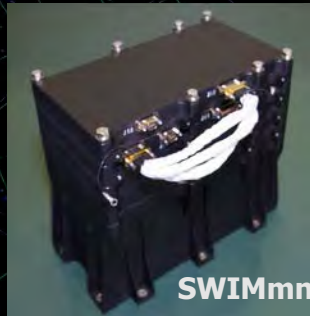
❖ Demonstration in orbit

SWIM on SDS-1

Space Cube 2 (SpC2) + Space Wire (SpW) system



SpC2



SWIMmn



Photo by JAXA

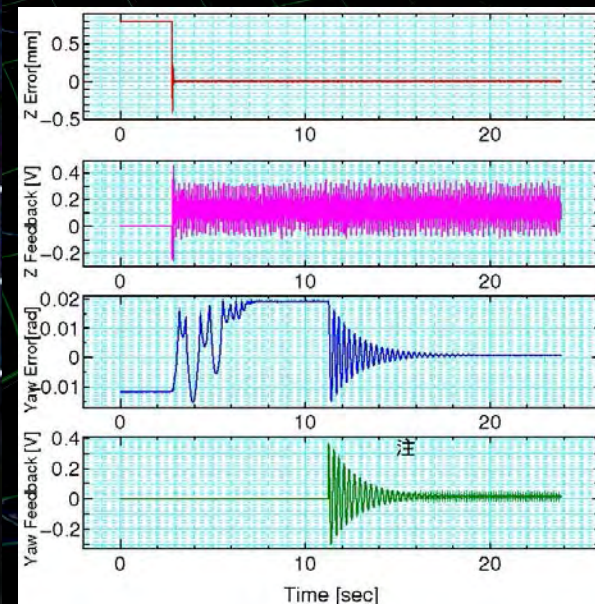
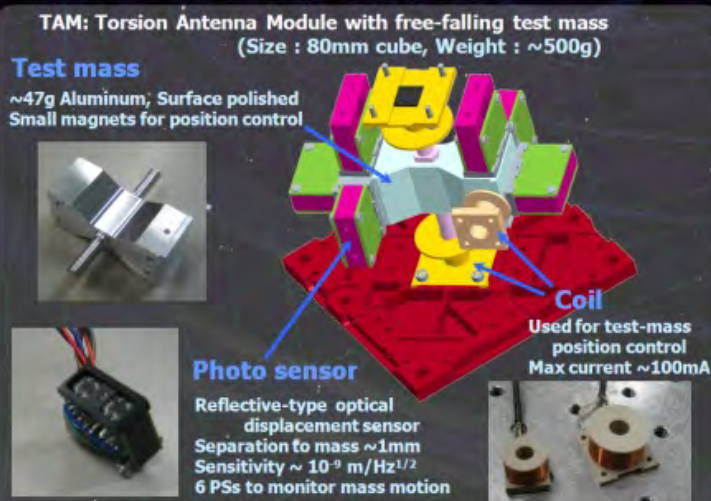
SDS-1
(Jan. 2009)

「いぶき」搭載カメラによる
衛星分離の様子
(2009年1月23日) ©JAXA

JAXA, U-Tokyo, Kyoto-U

❖ Control and precision measurement

SWIM demonstration in orbit (2009)



Mission status of DECIGO pathfinder

❖ JAXA's "Small science satellite series" program

*"At least 3 satellites in 5 years with
Standard Bus + M-V follow-on rocket"*

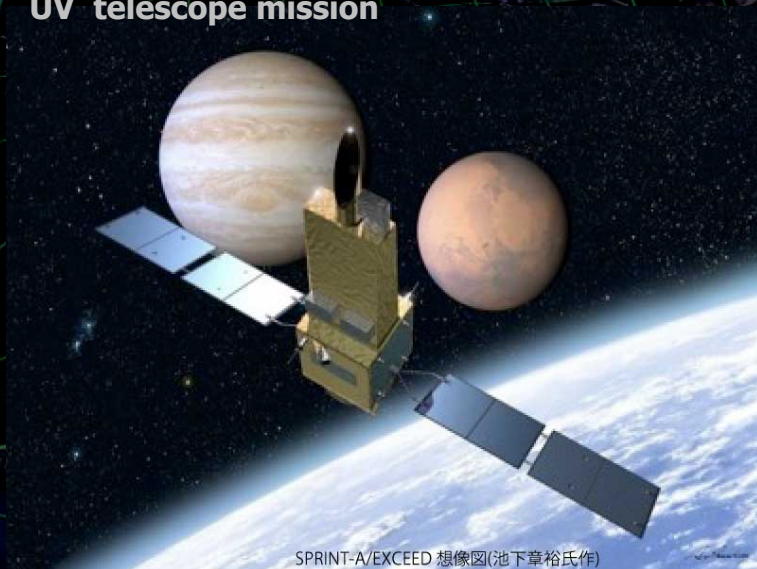
1st mission (2013): SPRINT-A /EXCEED

2nd mission (~2014/15) : SPRINT-B /ERG

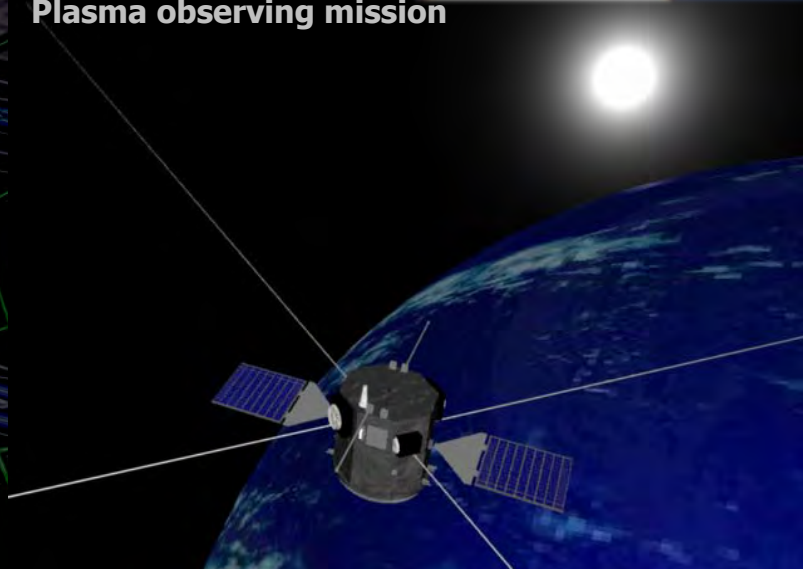
3rd mission (~2016/17) : SPRINT-C? /TBD? ←cancelled

Call for proposal : 2013?

SPRINT-A /EXCEED
UV telescope mission



SPRINT-B /ERG
Plasma observing mission



Epsilon rocket



Next-generation
Solid rocket booster (M-V FO)
Fig. by JAXA

Mission status of DECIGO pathfinder

- ❖ JAXA's "Small science satellite series" program

1st mission (2013): SPRINT-A /EXCEED

2nd mission (~2014/15) : SPRINT-B /ERG

- ❖ JAXA's "small science mission using epsilon rocket" program

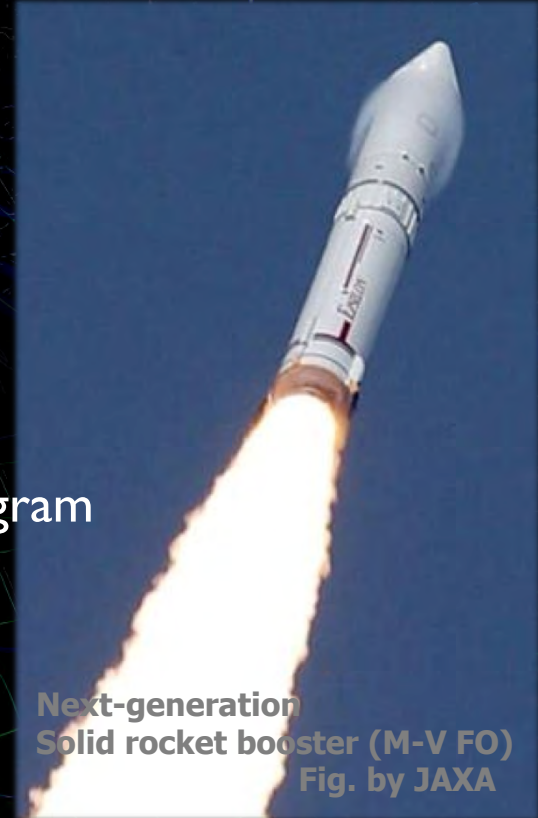
- ❖ The program framework was changed:

~10 M\$ payload mission to ~150 M\$ full mission

- ❖ AO from JAXA : December 2013

- ❖ Deadline : End of February.

- 7 mission proposals including DPF



Next-generation
Solid rocket booster (M-V FO)
Fig. by JAXA



DPF was dropped in the first down-selection (May)



Started discussions on the next strategy

Thinking back on DECIGO...

- ❖ DECIGO : Suggested in 2001
Seto, Kawamura and Nakamura, *PRL87, 221103 (2001)*
- ❖ DPF : Suggested in 2005
Ando, DECIGO workshop (2005)
- ❖ Good chance to reconstruct the strategy to DECIGO
 - ❖ Object of mission
 - ❖ Instrument design
 - ❖ Strategic demonstration of technologies

Summary

DECIGO : Rich Sciences !

Very beginning of the Universe

Dark energy, Dark matter

Galaxy formation

DECIGO Pathfinder

Mission proposal failed in selection

Good chance to reconstruct the strategy to DECIGO

Started discussions on the next strategy